

Prepared for: Ahart Aviation Services

186 Airway Blvd Livermore, Ca. 94551 tel 925-449-2142

fax925-373-0944

ahart@ahart.com

Ahart Aviation Seminole Checkout Test

Pr	epared by: Bill Komanetsky, Owner/Operator	
	rsion: 200810a	
D	ate:	
St	tudent/Customer Name:	
	(Please Print)	
Eı	ngines	
	What is the rated horsepower for each engine?	
2.	At what power setting and RPM will you attain this rated horsepower?	
3.	Engine cowlings are equipped with cowl flaps for increased engine cooling. Under what engine conditions will you open the cowl flaps?	
Pı	ropellers	
1.	Pitch of the propellers is controlled by engine oil and nitrogen pressure. Engine oil sends the propeller toward RPM. Nitrogen pressure and a large spring sends the propeller towards RPM.	
2.	What pressure should the nitrogen dome in the propeller hub be charged to?	
3.	If nitrogen gas is unavailable, what may be substituted when charging is necessary?	
4.	What reaction would you expect from the operation of the propeller if the nitrogen charge the propeller hub was insufficient?	
5.	If engine oil pressure were to drop to "0", what pitch setting would the propeller go to?	
6.	Will the propellers "feather" when parked on the ramp when the engines are not running? Why?	

Version: 200810a Page 1 of 10



tel 925-449-2142 fax925-373-0944

ahart@ahart.com

7.	. While airborne, you wish to save an engine by feathering the propeller. To do so you must move the propeller control to the feather position before the RPM drops below			
8.	Approximately how long will the propeller take to feather?			
9.	What steps would you take to un-feather a propeller?			
Lá	anding Gear			
	The landing gear is normally retracted by what source of power?			
	Approximately how long will it take for the landing gear to completely retract?			
	What keeps the landing gear in the retracted position?			
4.	What alternate way do you have of lowering the landing gear?			
5.	What is indicated by the illumination of the Red Gear Warning Light on the instrument panel?			
6.	Other than the 3 green down light, what additional way can you determine that the landing gear is down and locked?			
7.	What three conditions will activate the gear warning horn?			
8.	What prevents the gear from inadvertent retraction should the gear selector be placed in the UP position when the aircraft is on the ground?			
9.	Where may you obtain access to the landing gear pump and reservoir?			
10.	In daylight, what would cause the green landing gear DOWN lights not to be illuminated to full intensity?			
	Landing gear tires should be inflated to what pressure? Mains: Nose:			
12.	. At what speed may the gear be extended manually?			

Version: 200810a Page 2 of 10



3. Does gear movement affect CG? Explain.

186 Airway Blvd Livermore, Ca. 94551 tel 925-449-2142

fax925-373-0944

ahart@ahart.com

W	Weight and Balance				
1.	What is the maximum weight capacity of the baggage compartment?				
2.	Does fuel burn affect CG in this aircraft? Explain				

4. Given the following information, determine the CG using weight and arm values.

Arm	Weight	
85.87	2537	
	1263	
80.5	200	
80.5	180	
118.1	160	
142.8	75	
95	108 Gallons	
	85.87 80.5 80.5 118.1 142.8	

	4.1.	What is the gross weight of the aliciant?
	4.2.	What is the center of gravity?
	4.3.	What is the moment?
Fı	uel S	system
		Seminole has how many fuel cells?
		is the minimum octane fuel the plane can use?
3.		ally, fuel is supplied to the engines using the engine driven fuel pumps. What backup is available if either of these become inoperative?
4.	What	indications would you expect to see if an engine driven fuel pump were to fail?
5.		ne following conditions, describe how you would position the fuel selectors and electric ump switches:
	5.1.	Tank to engine (normal cruise flight).
	5.2.	Left engine inoperative, tank to engine.

Version: 200810a Page 3 of 10



tel 925-449-2142
fax925-373-0944
ahart@ahart.com

Temperature 38 C 28 C Headwind 5 Knots 5 Knots Elevation	ON during what preflight and		
How many fuel drains are available to check for fuel contamination during where are they located? Keoff Performance Given the following conditions at gross weight, please answer the question Livermore Airport (KLVK) South Lak Temperature 38 C Headwind 5 Knots 5 Knots Elevation	preflight and		
where are they located? Keoff Performance Given the following conditions at gross weight, please answer the question Livermore Airport (KLVK) Temperature 38 C Headwind 5 Knots 5 Knots Elevation			
Given the following conditions at gross weight, please answer the question Livermore Airport (KLVK) South Lak Temperature 38 C Headwind 5 Knots 5 Knots Elevation	ns below.		
Temperature 38 C 28 C Headwind 5 Knots 5 Knots Elevation			
Headwind 5 Knots 5 Knots Elevation	e Tahoe (KTVL)		
Elevation			
 What is the normal procedure Accelerate and stop distance at both airports? KVLK: KTVL:			

Version: 200810a Page 4 of 10



tel 925-449-2142

fax925-373-0944

ahart@ahart.com

Electrical System

1.	The electrical system is powered by what primary source?			
2.	What is the alternate or backup electrical power source?			
3.	What is the capacity and voltage of the aircraft battery?			
4.	What is the capacity of each alternator (in ampere-hours)?			
5.	If the alternator output exceeds approximately 17 volts, what would normally result?			
6.	Two voltage regulators maintain effective load sharing between both alternators while regulating electrical buss voltage. What is this voltage?			
7.	Approximately, what RPM is required to obtain full alternator output?			
8.				
9.	What is indicated if a significant reading remains on the ammeters after all the electrical equipment (except the master switch) is turned off?			
10.	What will normally result if an electrical unit powered by the main electrical buss were to experience a sudden surge of current, an electrical short circuit, or an electrical malfunction?			
	10.1. Can the failed system or unit be restored?			
	10.2. How can it be restored if it can be restored??			
11.	The ammeter for the left alternator reads "0". What may be done to return this alternator to normal operation?			
12.	How many times should the alternator circuit breaker be reset and why?			

Version: 200810a Page 5 of 10



tel 925-449-2142

fax925-373-0944

ahart@ahart.com

Vacuum System

1.	The vacuum system operates what instrument(s) in the aircraft?		
2.	The vacuum pressure system is regulated to provide vacuum pressure between what values?		
3.	How could vacuum pressure be increased if below normal values?		
4.	Will the capacity of the vacuum system be insufficient if one of the vacuum pumps were to fail? Explain why.		
Pi	tot/Static System		
1.	What instruments are operated by the Pitot Tube?		
2.	What instruments are operated by the static port?		
3.	Where does the alternate static source get its air?		
4.	What error would you expect to see when using the alternate static source?		
5.	The pitot heat should not be left on for longer than what time period before takeoff or landing?		
6.	Is pitot heat a de-icing or an anti-icing device?		
7.	If incorrect airspeed readings are being displayed during flight and there is no icing in or on the pitot tube, what could be the causes?		

Version: 200810a Page 6 of 10



tel 925-449-2142

fax925-373-0944

ahart@ahart.com

Heat, Ventilation and Defrosting Systems

1.	Heated air for the cabin and windshield defrosting is provided by what source?
2.	The cabin heater gets its fuel from what source?
3.	Describe the necessary position of each of the following controls to start the cabin heater
	3.1. Air Intake Lever:
	3.2. Temperature Control Lever:
	3.3. Defrost Control Lever:3.4. Fan/Heater Switch:
4.	Will the heater and/or fan operate with the Air Intake lever in the closed position?
5.	Will the ventilation blower operate during flight?
6.	How can the heating system be used to ventilate the cabin when heat is not desired?
7.	What is indicated by the illumination of the overheat light on the control panel?
8.	How can the heater be restarted after this light has been illuminated?
9.	What action must be taken to prevent the activation of the overheat switch upon normal shutdown during ground and flight operations? Ground:
	Flight:
10.	For flight planning purposes, what is the average fuel consumption of the heater?
	nergency Exit
	Where is the emergency exit located?
2.	How is the emergency exit operated?

Version: 200810a Page 7 of 10



tel 925-449-2142

fax925-373-0944

ahart@ahart.com

Cı	<i>r</i> uise	Performance
1.	Given	the following details, please answer the questions
		Altitude: 9,500 feet
		Temperature: Standard
		Power Setting: 55%
		What is the fuel burn per hour? Best Power:, Best Economy:
		What is the leaning procedure?
		est Power:
	Be	est Economy:
		What is the cruise range with a 45 minute fuel reserve?
		est Power:
		est Economy:
		What is the cruise endurance with a 45 minute fuel reserve?
		est Power:
	Be	est Economy:
La	ndir	ng Performance
		the following details, please answer the questions
		Temperature: KVLK: 38C, KTVL: 28C
		Wind Component: 5 knots at both KLVK and KTVL
		Aircraft Weight: 3800 pounds
	1.1.	What is the landing ground roll?
		_VK:, KTVL:
	1.2.	What is the landing distance over a 50 foot obstacle?
		, KTVL:
		What is the approach speed?
	KL	, KTVL:

Version: 200810a Page 8 of 10



tel 925-449-2142

fax925-373-0944

ahart@ahart.com

Emergency Procedures

1.	Durir –	ng single engine operations, when Vmc is reached, what is the recovery procedure?
2.	- Wha	t is the stall recovery procedure?
3.	- Wha	t is the procedure to identify then secure a failed engine in flight?
4.	- - Wha	t is the engine restart procedure during flight?
	- - -	
M		ellaneous t is the speed to properly latch the cabin door while in flight?
	Wha Wha follow	t is the maximum speed while using the autopilot? t altitude loss would you expect during a possible malfunction of the autopilot during the wing flight conditions? Climb: Descent: During an Approach:

Version: 200810a Page 9 of 10



tel 925-449-2142

fax925-373-0944

ahart@ahart.com

Aircraft Details

1. Fill in the following values

Vso		Total Fuel Capacity	
Vmc		Total Usable Fuel	
Vs1		Minimum Fuel Grade	
Vsse		Minimum Oil Capacity	
Vx		Maximum Oil Capacity	
Vxse		Normal Oil Capacity	
Vy		Oil Viscosity	
Vyse		Max Ramp Weight	
VIo (up)		Max Takeoff Weight	
Vlo (down)		Max Landing Weight	
Vfe		Max Baggage Weight	
Vno		Vacuum Min/Max	/
Va	3800 lbs:2700 lbs:	Maximum Demon- strated Cross Wind	

CFI Name:		
	(Please Print)	
CFI Signature:		

Version: 200810a Page 10 of 10